

# Carbon Fiber SMC

Charles Knakal

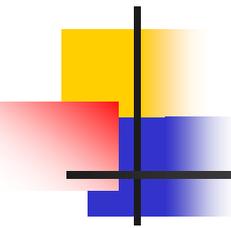
C. S. Wang

USCAR

General Motors

5-20-09

Project ID: Im\_07\_kia



# Overview

## Timeline

- Start – May, 2007
- Finish – Dec, 2010
- 25 % Complete

## Barriers

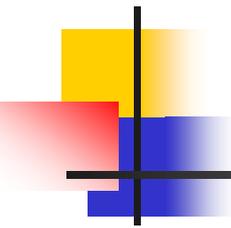
- Poor carbon fiber interface with automotive grade resin systems
- Poor mold flow
- Inconsistent material properties
- High costs

## Budget

- Total –
  - DOE: \$310,000 (including capital)
  - Contractors: \$60,000
- \$75,000 (plus \$110,000 ACC capital) in 2008
- \$54,000 for 2009

## Partners

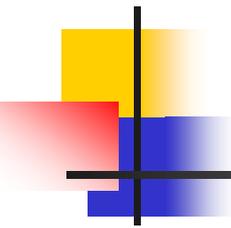
- Continental Structural Plastic (CSP), a Tier One supplier
  - Discounted compounding and molding
- Zoltek, a carbon fiber manufacturer
  - Discounted fibers



# Objectives

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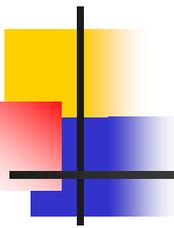
- To develop a carbon fiber reinforced SMC with physical properties and processing significantly superior to current carbon fiber SMC materials, in order to expand its application for automotive light-weighting.
  - > 200 MPa Tensile Stress
  - > 40 GPa Tensile Modulus
  - > 0.5% Tensile Strain to failure
  - < 10% COV
- To focus on lower cost carbon fibers and commercial viability.



# Approach

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- Work directly with a Tier One automotive SMC compounder/molder to develop an improved low cost carbon fiber SMC. Utilize the supplier's familiarity with existing material systems to rapidly implement the new technology.
- Investigate methods to improve carbon fiber distribution and fiber to resin matrix adhesion.
  - Fiber resin wet-out and adhesion
  - Molding consistency
- Develop structural SMC first, follow by Class-A material.

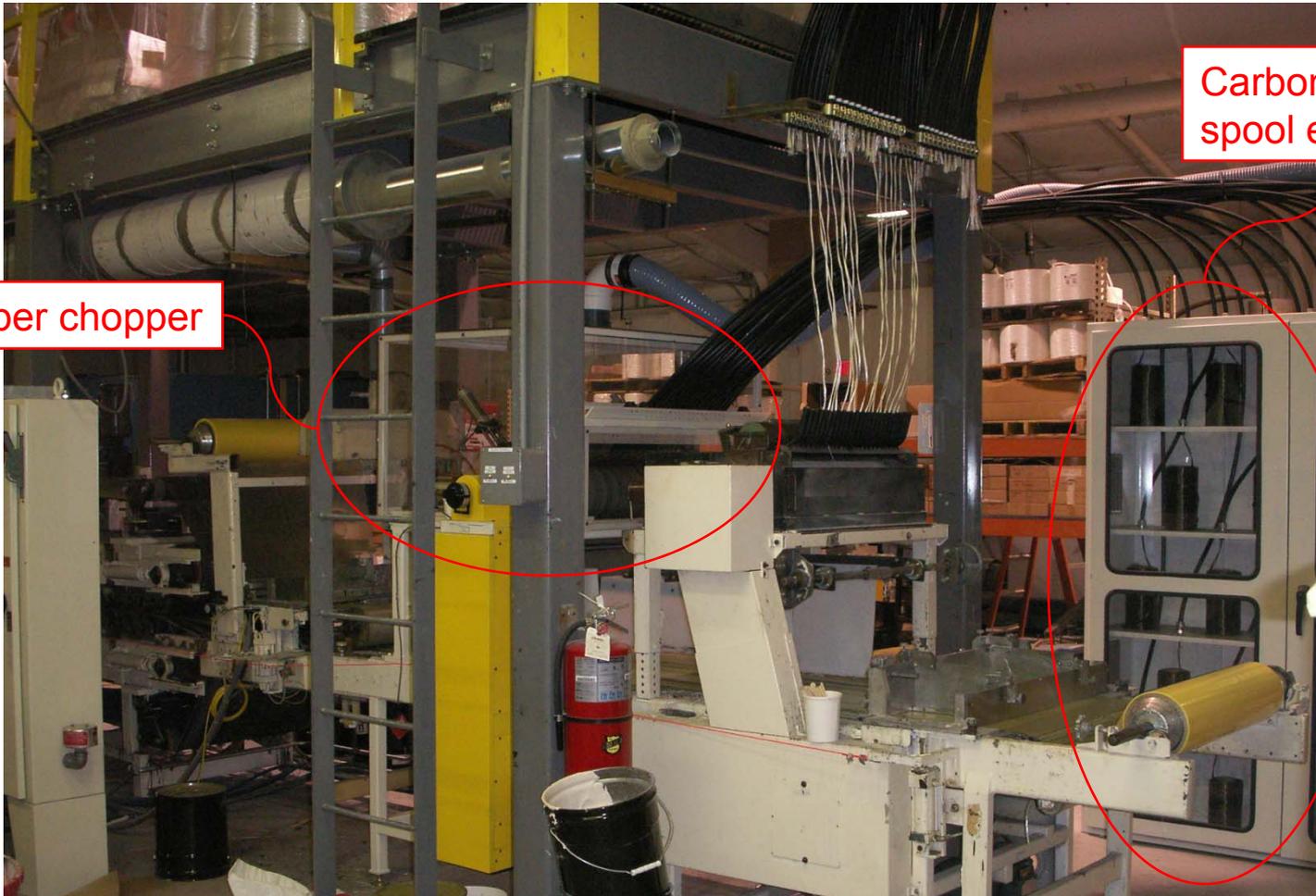


# Milestones

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- Secure Tier One supplier
- Modify SMC compounder
  - Funding
  - Modification
- Determine material goals
- Compound baseline material
- Develop improved material
- Mold demonstration automotive part

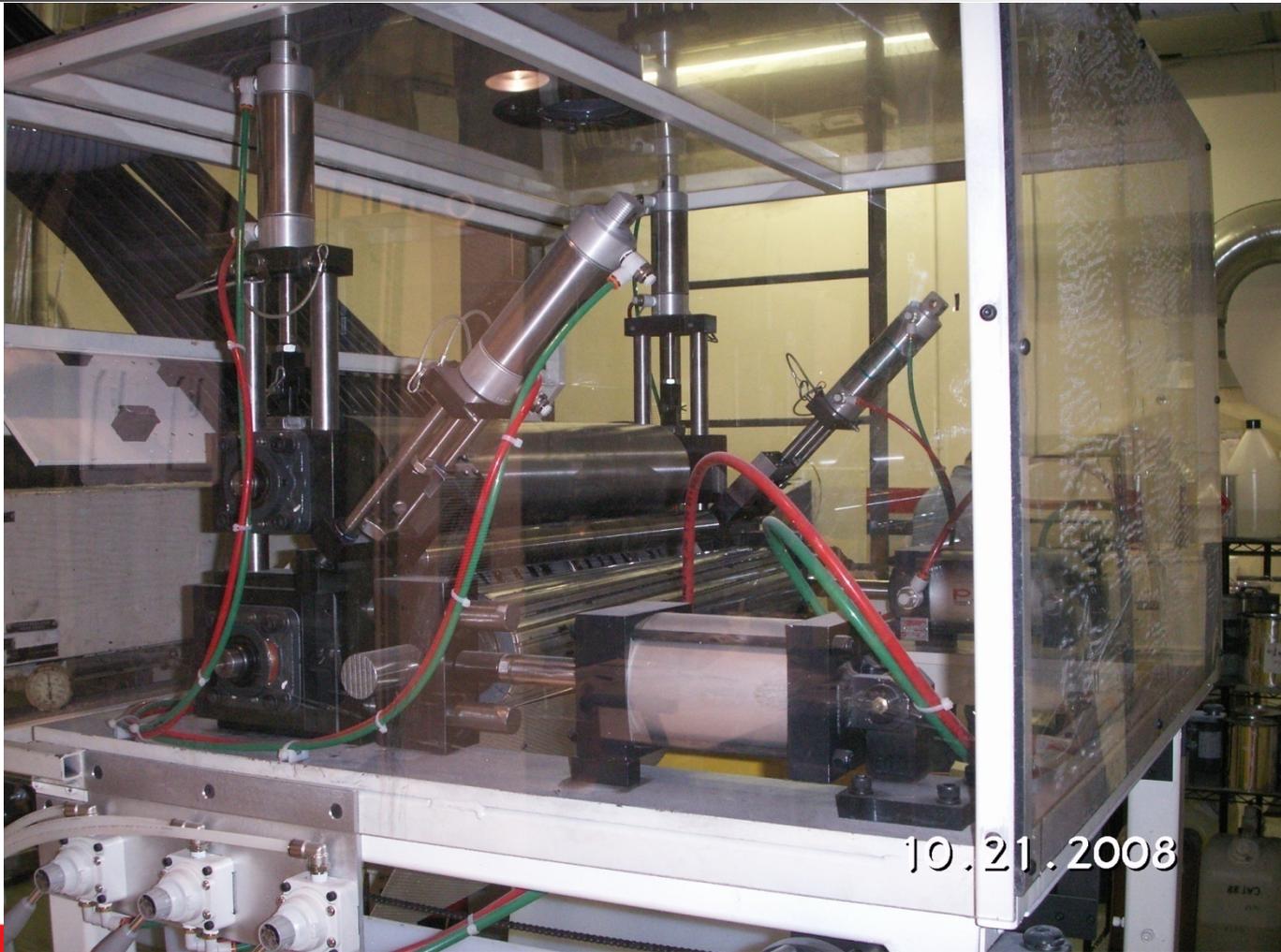
# SMC Compounder at CSP



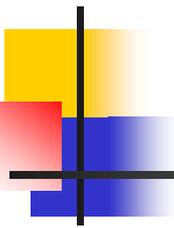
Carbon fiber chopper

Carbon fiber spool enclosure

# Carbon Fiber Chopper



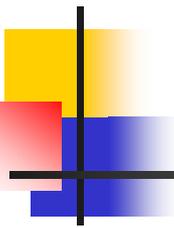
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# Technical Accomplishments – FY2008

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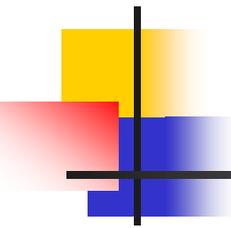
- Existing glass fiber SMC compounder was successfully modified to chop carbon fibers. This includes safety features for dealing with air born carbon fibers.
- Completed testing of currently available carbon fiber SMC materials, used to set project targets for an improved material.
- Conducted fiber “sizing” study of Panex 35 carbon fibers, determined best initial sizing and amount of sizing to use with automotive vinyl ester resin systems.



# Future Work

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- Compound and test baseline SMC material.
- Compound and test smaller tow carbon fibers to evaluate fiber distribution and wet out.
- Investigate methods to improve distribution and wet-out of low cost/large tow carbon fibers.
- Evaluate flow-ability of materials.
- Modify successful structural materials for improved surface appearance.
- Investigate/compound current “fast” epoxy SMC resin systems for comparison.



# Summary

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- Program targets were set to provide a commercially viable carbon fiber SMC material.
- CSP was selected as the Tier One supplier and their existing SMC compounder was upgraded to handle chopping of carbon fibers.
- An initial fiber sizing study was completed at Zoltek (the low cost carbon fiber supplier).

# Bond-Line Read-Through

Kedzie Fernholz

Ford

5-20-09

LM07

## Overview

### ■ Timeline

- Project Start: 3Q05
- Project End: 3Q10
- Percent Complete: 60%

### ■ Budget

- Total project funding
  - DOE \$600k
  - Contractor \$45k
- FY08 Funding: \$110K
- FY09 Funding: \$110K

### ■ Barriers Addressed

- Robust Joining Technologies for Composites
- Barriers to Implementation of Class “A” Carbon Composites
- Affordable Carbon Composites

### ■ Partners

- Visuol Technologies
- Meridian Automotive Systems (Experimental)
- Multimatic Engineering Services (Analytical)

# Objectives

## *Project Objective*

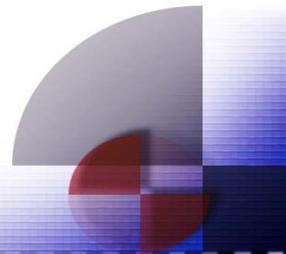
Develop the ability to predict bond-line read-through in the design phase to enable use of minimum thickness closure panels

## *FY08 Project Objectives*

- Phase 1 – Measurement Development
  - Evaluate and Refine Algorithm Converting Raw Data to Meaningful Quantitative Value
- Phase 2 – Determine BLRT Root Cause
  - FY08 Experiments
    - Initial Factor Screening Experiments
    - Effect of Cure Temperature Experiment
    - Effect of Adhesive Volume Experiments
    - Flange Coverage Experiment

# FY08 Milestones

- Phase 1 – Measurement Development
  - Demonstrate the developed measurement algorithm is applicable to experimental panels
  - Determine repeatability and reproducibility of the measurements.
  
- Phase 2 – Determine BLRT Root Cause
  - Identify factors with a high impact on BLRT severity
  - Identify at least two factors with a minimal impact on BLRT severity



## Approach

- Phase 1 – Measurement Development
  - Develop a measurement technique that quantifies the visual severity of surface distortions caused by bond-line read-through in a way that correlates with visual assessments
- Phase 2 – Determine BLRT Root Cause
  - Experimentally determine which material and process factors are the primary contributors to BLRT-induced distortions
  - Create experimental data to validate analytical models
- Phase 3 – Develop an Analytical Model for Predicting BLRT
  - Determine the material properties and analytical modeling techniques necessary to predict BLRT-induced distortions
  - Identify design principles to minimize the occurrence of BLRT and allow OEMs to use minimum thickness outer panels in closures

# FY08 Technical Accomplishments

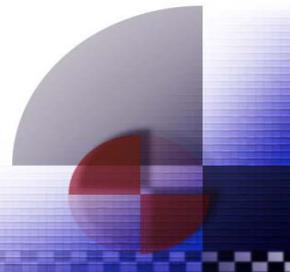
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## ■ Phase 1

- Algorithm demonstrated to successfully quantify BLRT distortions on experimental assemblies
- Repeatability & reproducibility of overall system found to be inadequate
  - Assemblies are measured three times to improve data
  - Evaluation of a more “production representative” system to occur in FY09

## ■ Phase 2

- Completed initial screening experiment, including analysis
- Completed initial follow-up experiments, including analysis
  - Effect of Cure Temperature Experiment
  - Effect of Adhesive Volume Experiments
  - Flange Coverage Experiment



# Phase 2: Experimental Analysis

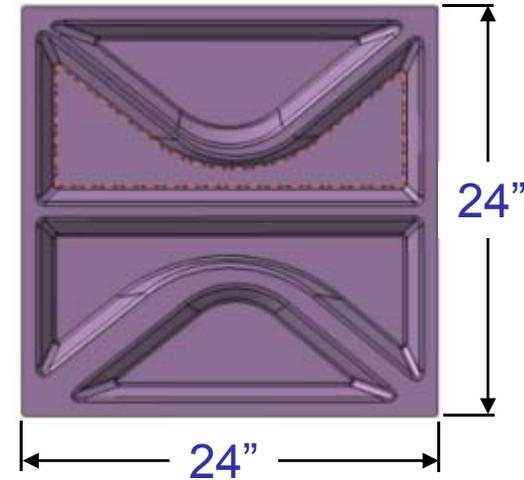
## Bond-line Read-through Samples

- Sample Geometry

- 24"x24" flat panel "outer panel"
- "Inner panel" tool with 4 flange widths

- Manufacturing Process

- Electrically heated bond nest
- Bond thickness controlled by bonding press
- Robotic application of adhesive

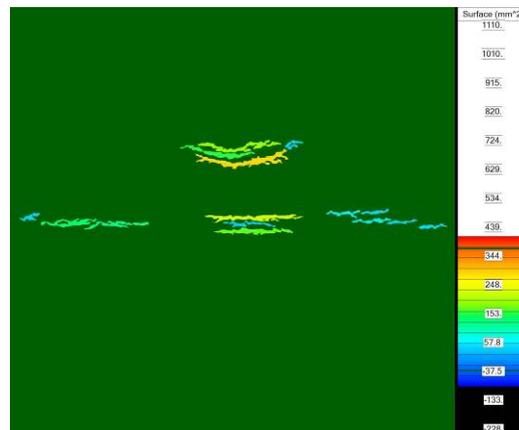


# Screening Experiment Results

- Stiffness (Modulus) of Outer Panel



2.5mm SMC



0.7mm Steel

*Bending (structural) stiffness is more important than Young's modulus!*

*The deflection of a plate is a function of the thickness cubed!*

$$U_{\text{plate}} = \frac{E t^3}{24 (1 - \nu^2)} \int \left\{ \frac{\delta^2 u_y}{\delta u_x^2} + \frac{\delta^2 u_y}{\delta u_z^2} \right\} dx dz$$

## Phase 2: Experimental Analysis

# Screening Experiment Results

- Type of bond nest had no effect
  - This may be due simply to the large percentage of the panel that is heated



“Full” Nest

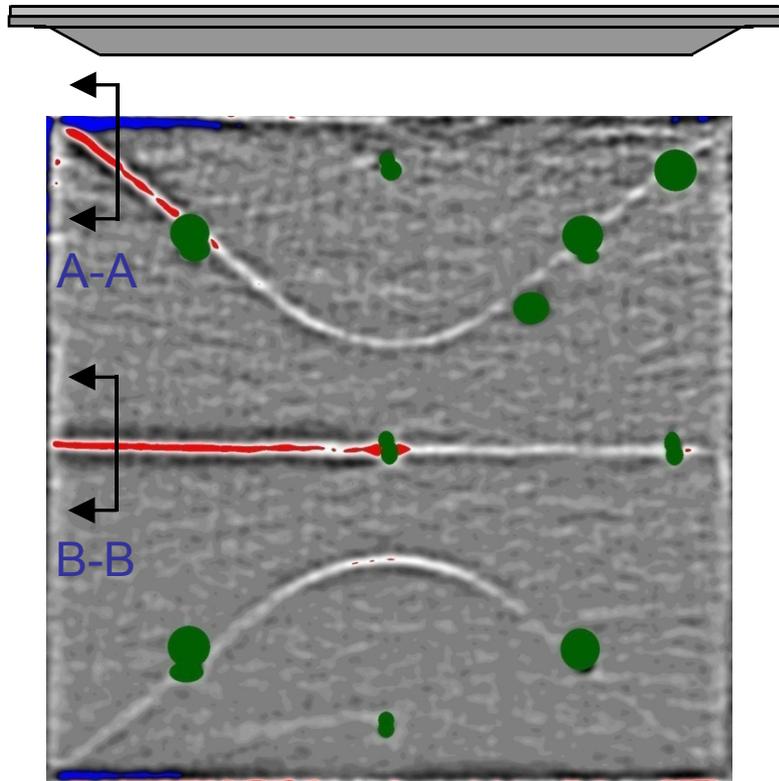


“Skeletal” Nest

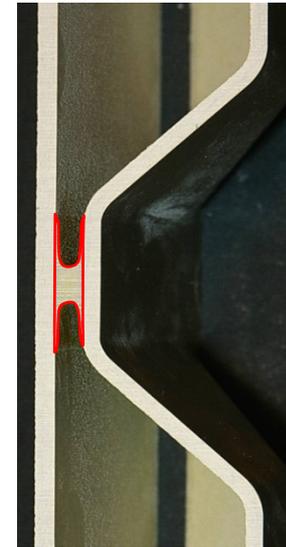


## Phase 2: Experimental Analysis

# Screening Experiment Results



Section A-A



Section B-B

*Note:* Section lines on curvature map are approximate locations

## Phase 2: Experimental Analysis

# Drop Size Evaluation

- Adhesives

- Epoxy
- Urethane

- Drop Sizes

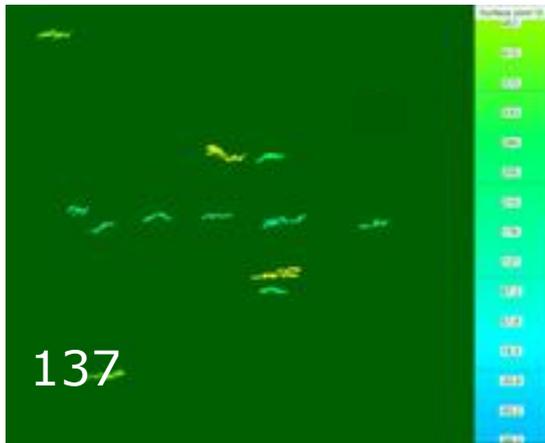
- Robotic Application Volume
- $\frac{1}{2}$  Robotic Application Volume
- $\frac{1}{4}$  Robotic Application Volume

Hand Dispensed Using a Syringe



# Drop Size Evaluation

## Urethane



Robotic  
Dispense

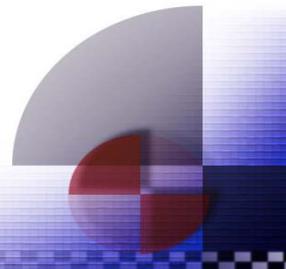


$\frac{1}{2}$  Robotic  
Dispense



$\frac{1}{4}$  Robotic  
Dispense

*Making the drops smaller eliminated BLRT.*



# Drop Size Evaluation

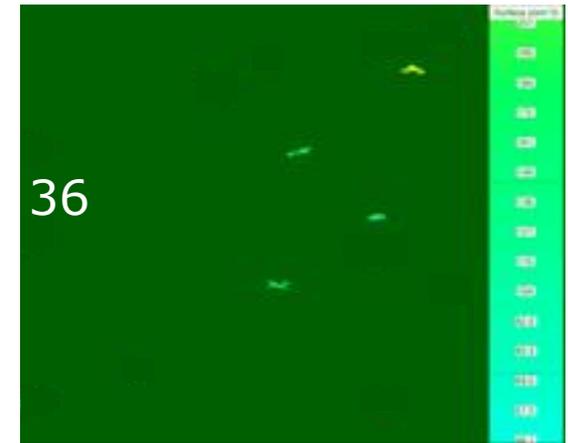
## Epoxy



Robotic  
Dispense



1/2 Robotic  
Dispense

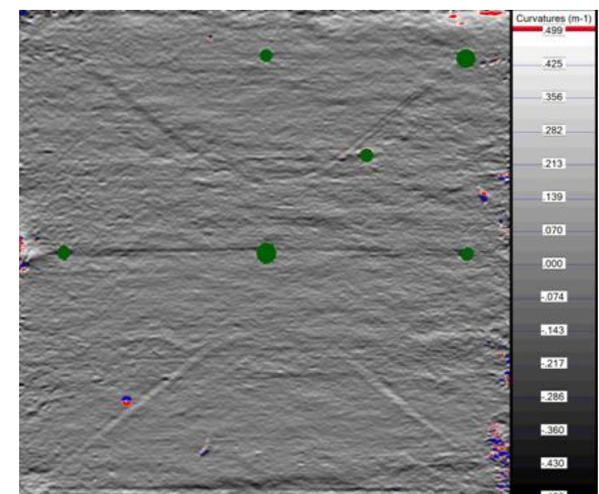
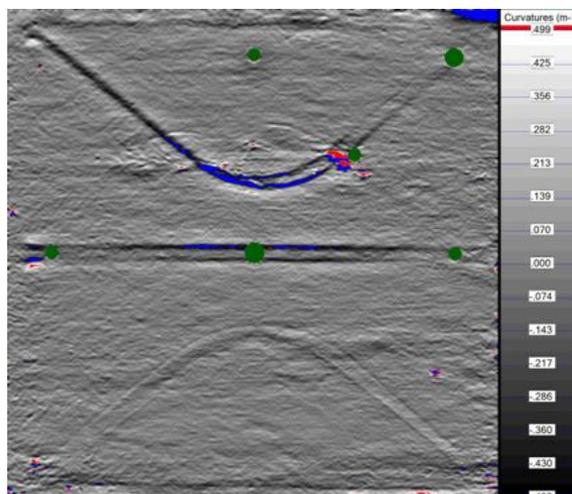
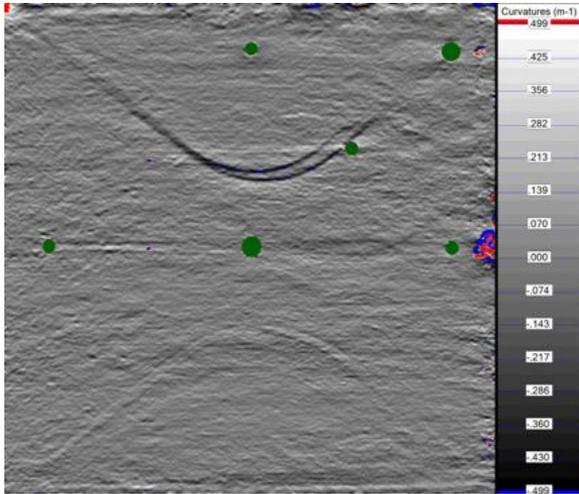


1/4 Robotic  
Dispense

*Making the drops smaller reduced BLRT.*

***Is this due to volume or squeeze-out?***

# Squeeze-out vs. Adhesive Volume



“Standard Dispense”  
1mm nominal bond gap

“2X Dispense”  
1mm nominal bond gap

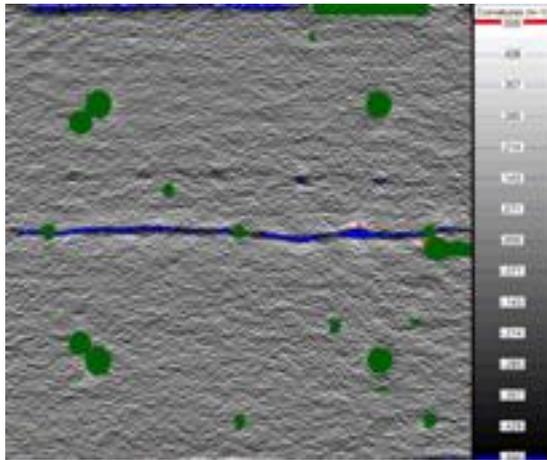
“2X Dispense”  
3mm nominal bond gap

***Squeeze-out  
appears to be more of a concern than the volume of adhesive.***



# Freestanding Outers – Epoxy

- Apply Adhesive to an Outer Panel
    - Bead Across the Center
    - Large Drops
    - Small Drops
- } Hand Dispensed Using a Syringe
- Cured in the fixture (300°F) until dry



*Adhesive causes more distortion on a freestanding outer panel than on an assembly!*

## Future Work

- Evaluate Variations on Measurement System Hardware to Improve Repeatability & Reproducibility
- Complete Additional Experiments
  - Mastic screening
  - Panel density and inner panel thickness
  - Generate experimental data to validate CAE model development
- Begin Development of Analytical Models
  - Develop a validated model for BLRT on a freestanding outer
  - Develop a validated model for BLRT on a “Basic” Assembly
  - Develop a validated model for BLRT caused by Stand-offs on the inner panel bond flange
  - Use the model to explore the effectiveness of different design strategies

## Summary

- Surface distortions caused by BLRT can now be quantitatively measured
- Experimental data generated in this project has identified several key material and process factors for which analytical models must account
- The ability to predict BLRT will allow OEMs to immediately reduce closure outer panel thickness (and therefore WEIGHT) by 25%.
- This technology will allow the use of minimum thickness panels when Class “A” carbon fiber SMC becomes technically and financially viable.

