



CSP Program Summit 2016

Direct s-CO₂ Receiver Development

Laboratory Proposal Development
Process (LPDP) FY2012-2015

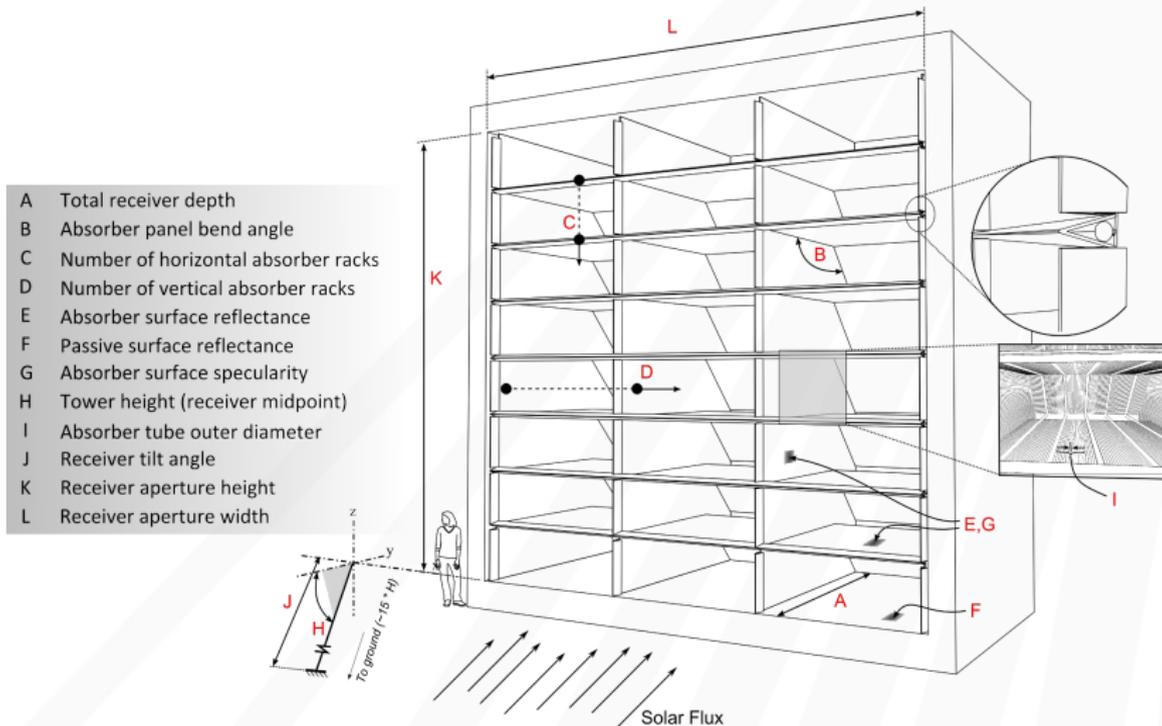
energy.gov/sunshot

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Design objectives and features

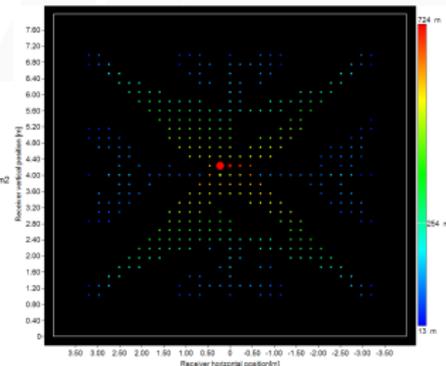
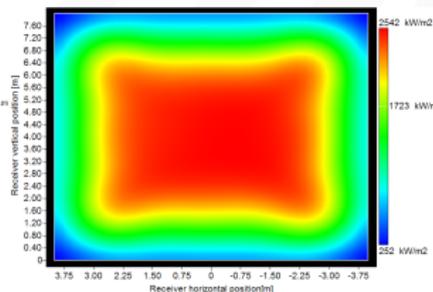
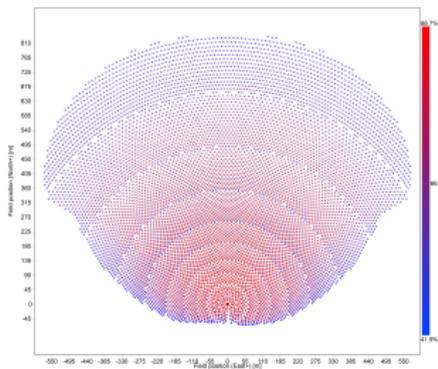
- Develop, characterize, and experimentally demonstrate a novel high-temperature direct s-CO₂ receiver
- Operate with >90% thermal efficiency at 650°C outlet temperature, 10,000 thermal cycles
- Unique use of geometry to improve performance and lifetime, reduce cost
- Use modeling techniques and tools help understand and optimize the entire technology
- Absorber is a tubular panel set arranged in a gridded geometry
- Absorber is partially absorptive to encourage flux scattering
- Heat shields protect header piping near aperture from incident flux
- Flux intensity at aperture is significantly higher than on absorbing surfaces

Receiver Concept

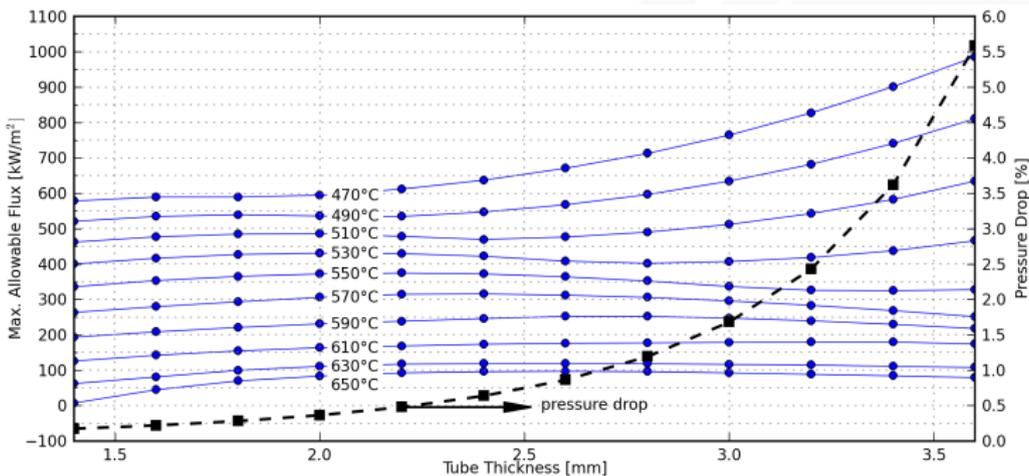


Solar Field Design

- Solar field is north-based for cavity configuration
- Flux aiming on receiver balances uniformity and spillage loss.



Maximum allowable flux drives receiver design

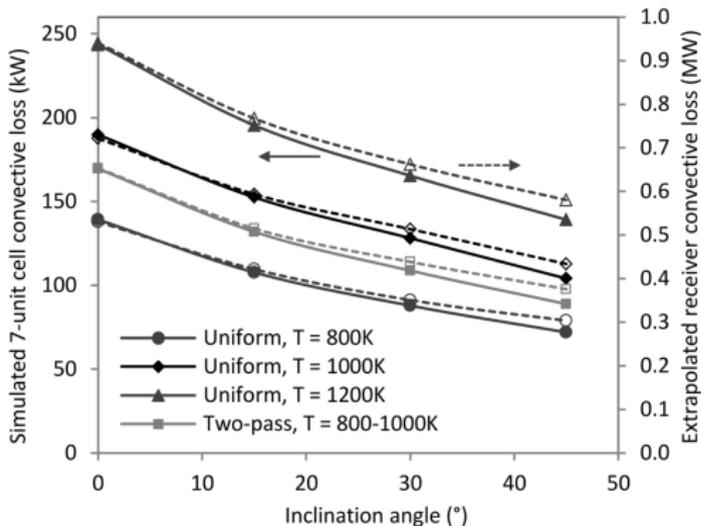


- Creep fatigue at inner surface is normally the limiting factor for high-pressure, high-temperature
- An optimized thickness balances max flux, pressure loss, cost

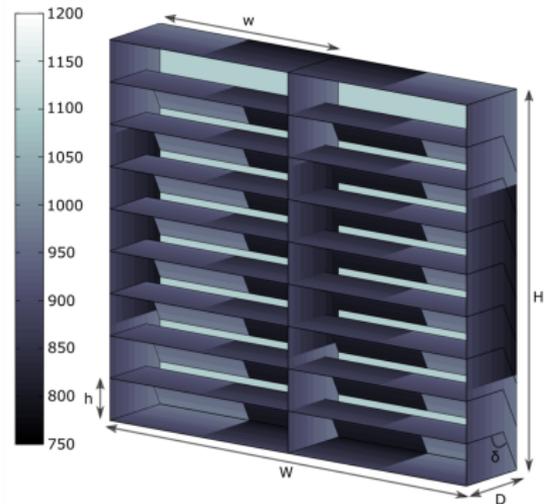
Natural convection modeling

- 3D Computational fluid dynamics (CFD) using ANSYS Fluent
 - SST k- ω turbulence model
- Full-receiver convective loss < 0.7 MW (**0.7%**) for 1000K wall temperature
< 1.0 MW (**1.0%**) for 1200 K wall temperature

Receiver convective loss



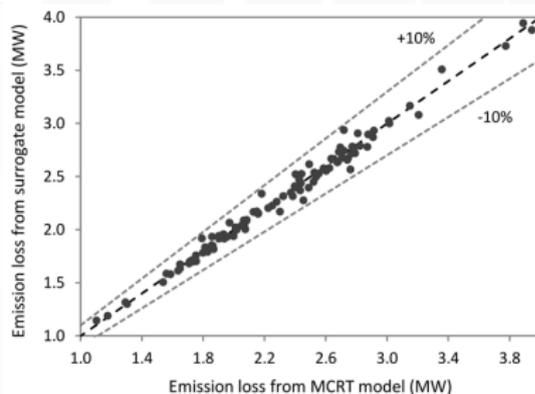
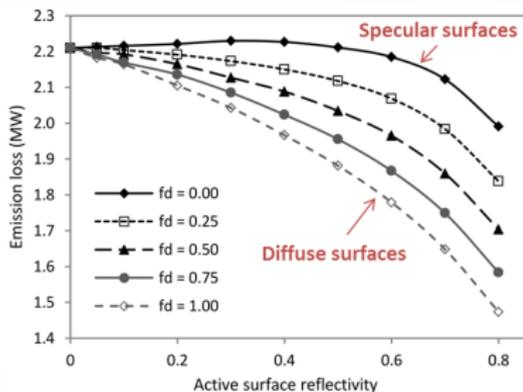
Two-pass wall T profile



Emission loss modeling

- Design was characterized over a broad design space
- Emission loss from the base configuration is between 1.5 - 2.2 MW
- Loss can be modeled using 13-dimensional regression

$$y = \beta_0 + \sum_{i=1}^N \beta_i x_i + \sum_{i=1}^{N-1} \sum_{j=i+1}^N \beta_{ij} x_i x_j + \sum_{i=1}^N \beta_{ii} x_i^2$$



Efficiency Definition and Measurement

Definition

Thermal efficiency is defined as heat absorbed by the s-CO₂ fluid divided by total heat incident on the aperture of the receiver

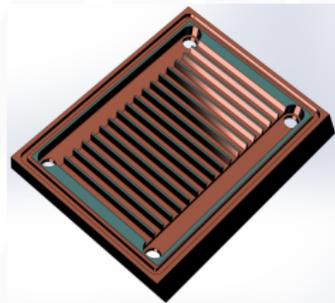
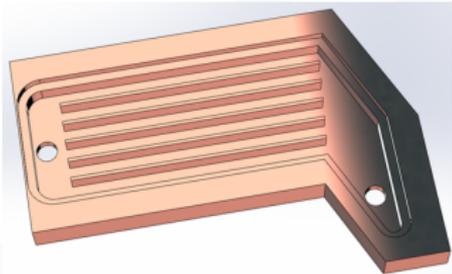
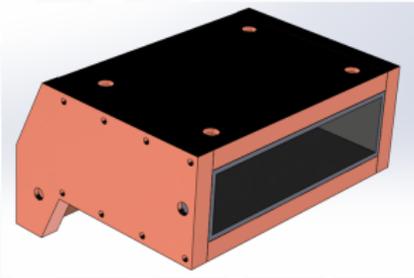
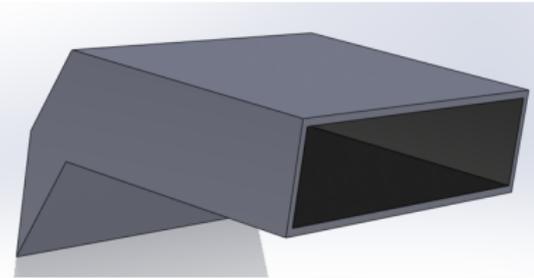
$$\eta_{therm} = \frac{\dot{m}_{CO_2}(h_{out} - h_{in})}{\int_x \int_y q''_{in}(x, y) dx dy} \quad (1)$$

Equivalent form in terms of heat loss:

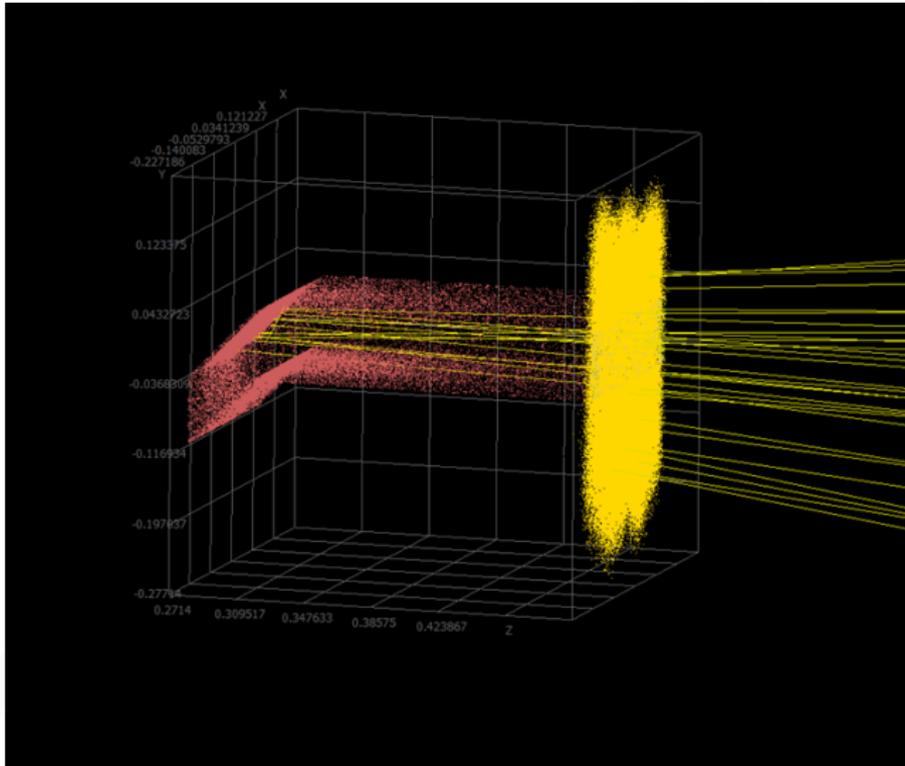
$$\eta_{therm} = 1 - \frac{1}{\int_x \int_y q''_{in}(x, y) dx dy} \left(\overbrace{\int_x \int_y q''_{out}(x, y) dx dy}^{\dot{q}_{refl}} + \underbrace{\sum_i \gamma(\theta) A_i (\bar{T}_i - T_\infty)}_{\dot{q}_{conv}} + \underbrace{\sum_i \sigma \epsilon A_i F_i (\bar{T}_i^4 - T_{amb}^4)}_{\dot{q}_{emit}} \right) \quad (2)$$

Model validation with bench-scale experiments

Calorimetric experiment measures heat absorption by unit cell



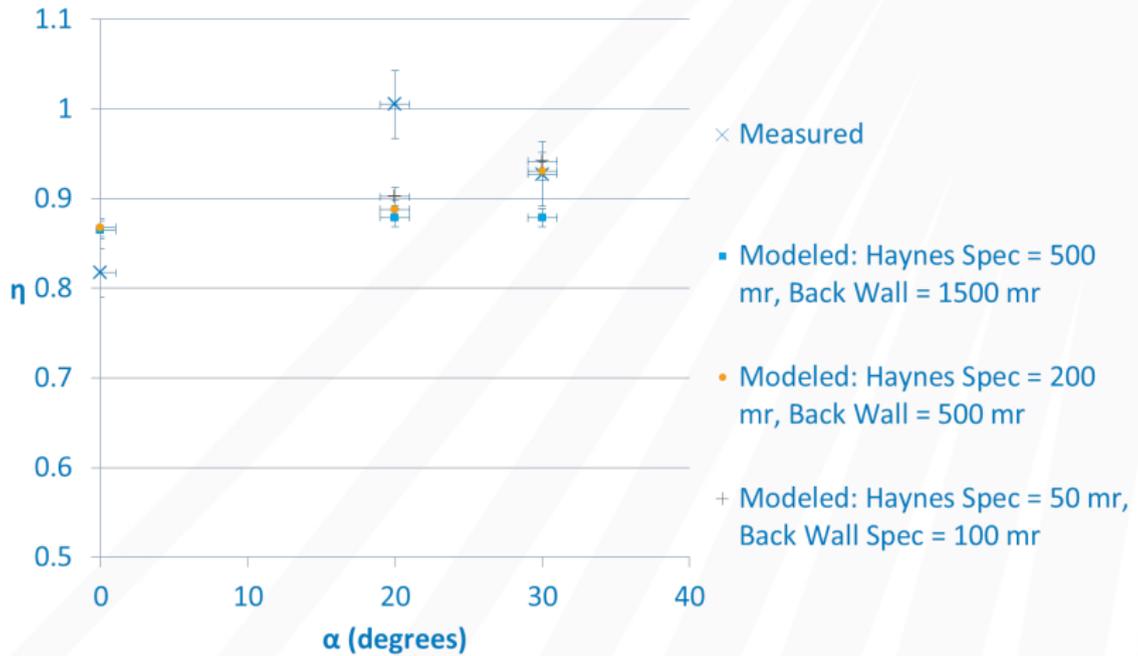
Tested at NREL's High Flux Solar Furnace



SolTrace ray trace of test cell at direct orientation ($\alpha = 0^\circ$, $\theta = 0^\circ$)

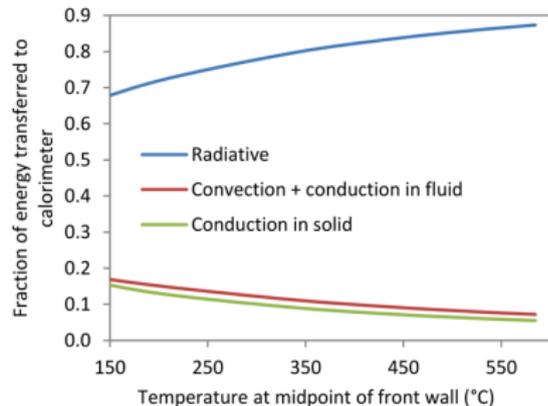
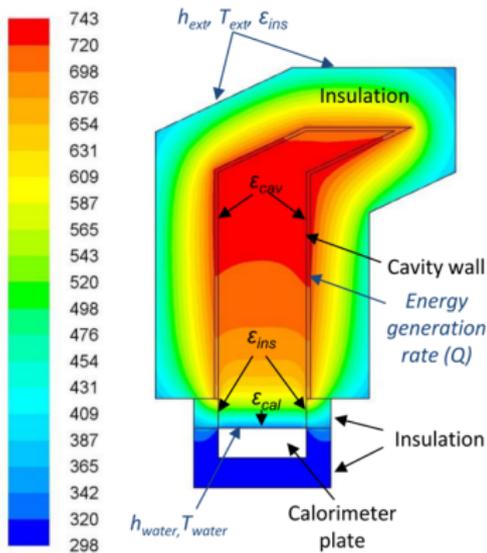
Experimental vs. model results: tilt angle subtest

Test Cell η vs. Tilt Angle α @ fixed Skew Angle $\theta = 0^\circ$

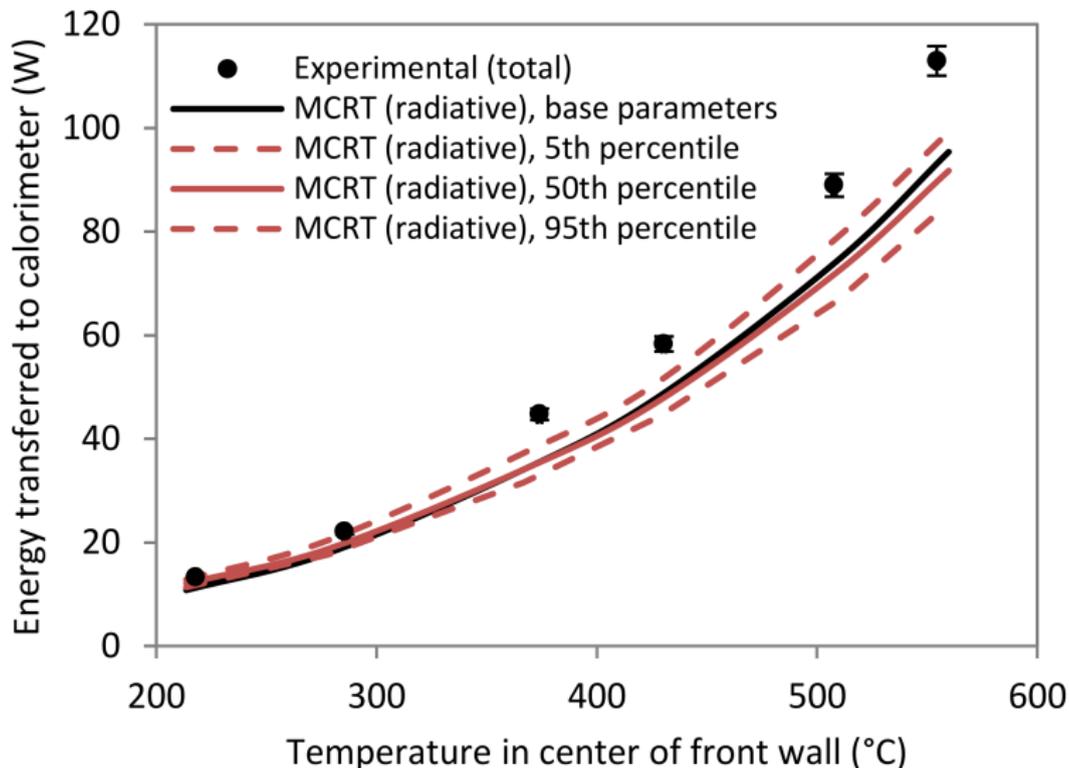


Emission loss model validation

- Analysis of full test setup using CFD / surface exchange model
- **70 - 90%** of energy transferred to calorimeter occurs by thermal radiation



Measured emissive heat transfer



Design Outcomes

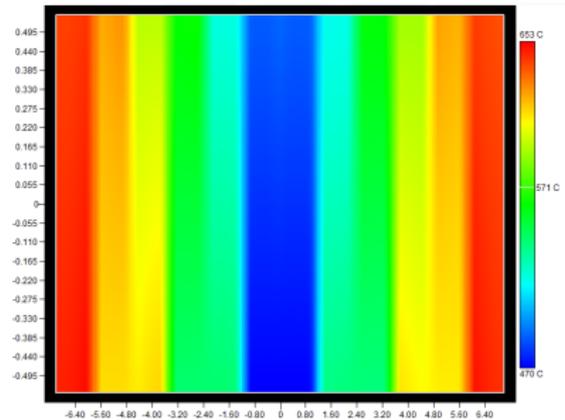
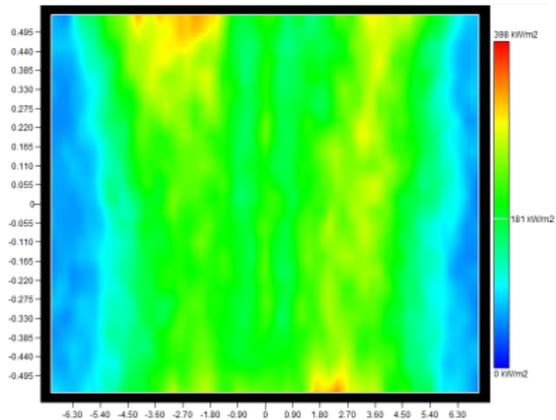
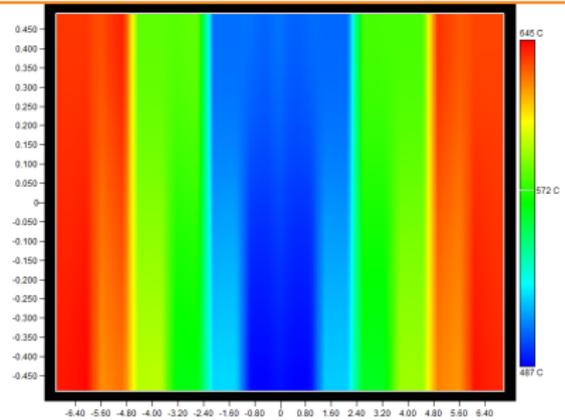
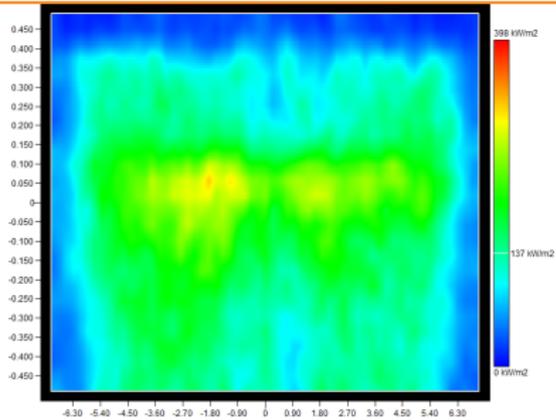
Design knowledge

- Cavity receivers have the potential to thermally out-perform external receivers
- Convective loss reduced by limiting vertical surfaces or using baffles or horizontal panels
- Receiver pressure loss significantly reduces cycle conversion efficiency of the power cycle
- Emissive and reflective loss are the most significant thermal losses this geometry
- Thermal absorption efficiency in a cavity system is only partially dependent on absorber surface reflectivity
- Tubing material usage is reduced by illuminating both sides of the panel
- Material stress drives the receiver design
 - Pressure, circumferential, radial, and gravity stresses can be significant
 - Allowable stress depends on creep, cyclical factors
 - *Data* on alloy properties are critical
- Metallurgical advances are critical for high-temperature development

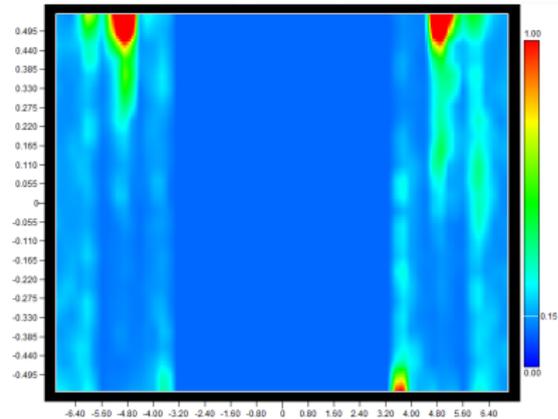
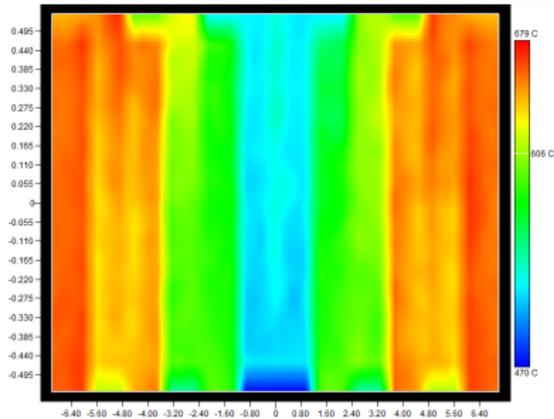
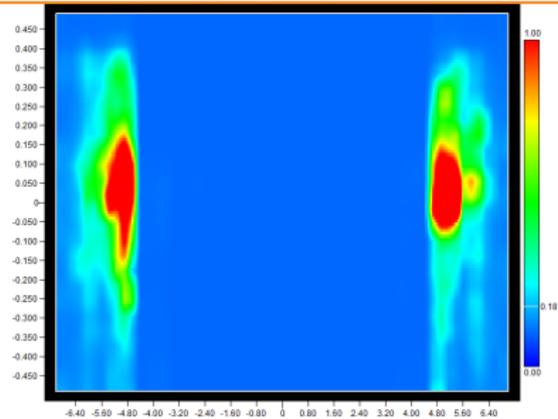
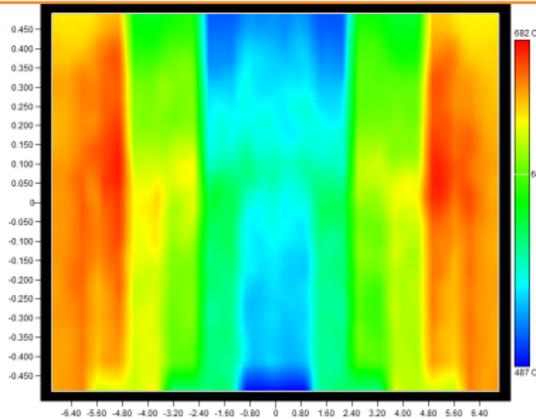
Design challenges

- High-temperature optical coating
- Pressure loss through the receiver, main header piping
- Allowable surface flux as a function of HTF temperature
- Front-edge (header?) shielding for the horizontal panels
- Aperture flux distribution, solar field requirements
- Flow rate control for parallel paths
- Design optimization and high degree of dimensionality
- Cost-effective thermal storage, cycle integration
- Heat shield reflective materials
- Flux over limit due to serpentine flow path
- Shallow cavity depth to reduce gravitational moments

Design Example



Design Example



Project Achievements

- I. Identified and demonstrated receiver concept that can achieve high efficiency, $> 92\%$ at design
- II. Developed modeling tools and methodologies that advance receiver design
- III. Experimentally validated thermal performance models
- IV. Improved understanding of allowable flux, stress, and creep
- V. NREL submitted provisional patent application on receiver concept