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# Recearch & Technology

Compressed Hydrogen Storage Workshop *Manufacturing Perspective* 

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#### **DOE Hydrogen Program**

Development of Advanced Manufacturing Technologies for Low Cost Hydrogen Storage Vessels

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Materials & Fabrication Technology

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Project ID #MF008

This presentation does not contain any proprietary, confidential, or otherwise restricted information







# **Timeline**

- Project start date 09/2008
- Project end date: 09/2011
- Percent complete: 60%

# **Budget**

- Total Budget: \$5,486,848
- DOE Share: \$2,566,451
- QT/Boeing Share: \$1,920,397
- FFRDC Share: \$1,000,000

# **Barriers**

- Material system costs
- Manufacturing processes

# **Partners**

- Quantum Technologies, Inc.
- The Boeing Company (Boeing)
- Pacific Northwest National Laboratory (PNNL)
- Lawrence Livermore National Laboratory (LLNL)

## Briefly – Composites at Boeing, 787 Family

BOEING



787-8 210-250 passengers 7,650-8,200 nmi (14,200-15,200 km)

787-9 250-290 passengers 8,000-8,500 nmi (14,800-15,700 km)

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### **Composite Structure**



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## Partners Across the Globe are Bringing the 787 Together

Engineering, Operations & Technology | Boeing Research & Technology Materials & Fabrication Technology U.S. Australia Asia Europe Fuji Boeing Boeing **Messier-Dowty** Spirit **Rolls-Royce** Mitsubishi Wing tips GE Canada Latécoère Kawasaki Seoul, Korea Goodrich Boeina **KAL-ASD** Alenia **Messier-Dowty** Saab **Fixed trailing edge** Wing Nacelles Nagoya, Japan Nagoya, Japan Mid forward fuselage Chula Vista, CA Nagoya, Japan Forward fuselage Moveable trailing edge Wichita, KS Center fuselage Melbourne, Australia Grottaglie, Italy 43 ······ Tail fin Cargo access doors Frederickson, WA Passenger entry doors Linköping, Sweden Toulouse, France Wing/body fairing 47 Landing gear doors 48 .... Winnipeg, Canada 11 45 Main landing gear Horizontal stabilizer wheel well Aft fuselage Nagoya, Japan Foggia, Italy Charleston, SC Engines Center wing box. **GE** – Evandale, Ohio Nagoya, Japan Rolls Royce - Derby, UK

> Landing gear Gloucester, UK

Fixed and moveable

leading edge

Tulsa, OK



#### 787-size airplanes represent 3,400+ market

Source: Boeing CMO 2010 - 2029

EOT\_RT\_Sub\_Template.ppt | 1/12/2009 | Structural Tech 7



## **Significant Need for Carbon Fibers in Aerospace**



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## ... Also for other Energy and Industrial Uses

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## **Specific Strength and Modulus**

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### **DOE High Priority MR&D Needs**



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### Table 10: Summary of High-Priority Manufacturing R&D Needs: Hydrogen Storage

Develop process technologies for reducing the cost of carbon fiber

Currently, composite tanks require highstrength fiber made from carbon-fiber grade polyacrylonitrile precursor. The price of the carbon fiber is typically about \$20/kg. Reducing the cost of the fiber by about 60%, or about \$6/kg, would yield significant savings in the unit cost of composite tanks. Manufacturing R&D is needed to develop lower cost, lower energy decomposition process for carbon fibers, such as microwave or plasma processing.

#### Develop new manufacturing methods for high-pressure composite tanks

New manufacturing methods are needed that can speed up the cycle time, that is, the per unit fabrication time. Potential advances in manufacturing technologies include faster filament winding (e.g., multiple heads), new filament winding strategies and equipment, continuous versus batch processing (e.g., pultrusion process). New manufacturing processes for applying the resin matrix, including tow-pregs for room temperature curing, wet winding processes, and fiber imbedded thermoplastics for hot wet winding, should also be investigated.

#### Develop manufacturing technologies for conformable high-pressure storage systems

Although this is a design issue (improved energy density), new manufacturing methods for carbon fiber winding and fiber placement manufacturing could also be applied to improve conformability of tanks by allowing modified cylindrical tank shapes to be manufactured.

#### Improve fiber placement processes

Fiber placement technologies can reduce unit costs by reducing the amount of carbon fiber needed by as much as 20%-30%. This approach may also allow some improvement in conformability of high pressure tanks. However, the process is slow. New methods and equipment are needed to improve manufacturing cycle time.

### **Hybrid Process**



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#### **Fiber Placement**





Filament winding process is not optimal

- Allows no easy dropping/picking-up of tows
- Build up of thickness at dome (ends) must pass across length of cylinder
- Results in 15 to 20% added weight
- Lower quality laminate higher porosity, lower fiber volume
- Fiber placement process is not optimal
- Slow process, 2 lb/hr lay down, compared to 30 lb/hr for filament winding
- Expensive equipment
- More touch labor
- Goal to manufacture Type IV H<sub>2</sub> storage pressure vessels, utilizing a new hybrid process with the following features
- Optimal elements of flexible fiber placement & commercial filament winding

With the aim of achieving:

 A manufacturing process with lower composite material usage, cost & higher efficiency

## Fiber Placement Technology

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- Fiber placement, a CNC process lays multiple strips of composite material on demand.
  - Allows maximum weight efficiency
  - Only places material where it is needed
  - Steering of fiber allows greater design flexibility

Ref: Boeing Released, BOE031709-109, by K. M. Nelson on Jan 29, 2010.

- Existing machines don't meet the all objectives
- Process is scalable to smaller parts
- Software available for smaller machines









### **Microstructure**



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# First Tank Fabrication and Test



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The smallest polar opening AFP can make currently

The regions (domes) covered by the localized reinforcement were protected very well.

- Static Burst Result: 23420 PSI > 22804 PSI, EN standard (New European Standard superseding EIHP)
- 64.9 kg composite usage in the 1st hybrid vessel vs.
   76.0 kg in the baseline tank (FW alone)
   11.1 kg Savings!



Quantum and Boeing's manufacturing experience was used to estimate the \$/kg of Filament Wound (FW) and Automatic Fiber Placed (AFP) Composites.

Hybrid composite design provided the mass of Filament Wound and Automatic Fiber Placed Composites.
Cost model included materials, labor, overheads, balance of system, manufacturing equipment and factory space costs.
Baseline and two bounding manufacturing scenarios were investigated:

- **1.** Baseline = Quantum Filament Wound 129 Liter, Type IV Tank.
- 2. <u>Fully Integrated FW and AFP</u> Composite layup optimized for high strength, but inefficient machine usage.
- 3. <u>Fully Separate FW and AFP</u> 100% machine usage, but composite strength may be slightly reduce.



# Tank Cost Analysis, 500,000/yr, \$11/lb Carbon Fiber



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Ref: Boeing Released, BOE012010-035, by K. M. Nelson on Jan 29, 2010.

Summary Table		Type IV Tank	Hybrid FW + AFP Reinforced	
		Baseline 129L	Fully Integrated	Separate
		Filament Wound	FW and AFP	FW and AFP
Composite Mass, kg	FW	76	63.4	63.4
	AFP		1.5	1.5
Total Composite Mass, kg		76	64.9	64.9
# Manuf. Cells for 500K/yr	FW	191	242	159
	AFP		484	165
Tank Costs				
FW Composite		\$2,290	\$1,910	\$1,910
AFP Composite			\$90	\$90
End Boss		\$250	\$250	\$250
Manufacturing Equipment		\$36	\$66	\$41
Factory Space		\$7	\$10	\$7
Total Tank Cost		\$2,583	\$2,326	\$2,299
% Tank Cost Savings		0%	10%	11%
DOE Measures				
Specific Energy, kWh/kg <sup>1</sup>		1.50	1.67	1.67
Cost Efficiency, \$/kWh <sup>2</sup>		\$23.45	\$21.91	\$21.75
<sup>1</sup> 5 kg H2 * 33.31 kWh/kgH2/(				ompMass=30kg
<sup>2</sup> (Tank+OtherComponents \$\$	5) / (5 k	g H2 * 33.31 kWh/k	gH2)	

\*Cost numbers provided by Pacific Northwest National Labs. See Notes.

## **Cost\* and Tank Weight Compared to DOE Targets**



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\*Cost numbers provided by Pacific Northwest National Labs. See Notes.



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Total Cost =\$3864 Tank Hardware System Hardware 500,000 Units per Year

#### \*Cost Numbers Derived from PNNL Cost Analysis. See Notes.



**Breakdown of System Costs** 







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### **DOE Energy Storage Goals**

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The current (as of 2009) high pressure, i.e. 700 bar (10,000 psi) gaseous, hydrogen storage

- Goal is a 5-kg H<sub>2</sub> storage vessel
- 2010 target for on-board hydrogen storage is:
  - \$4/kWh by 2010
  - 2 kWh/kg (6 wt%)
  - 1.5 kWh/L
- By 2015 target is:
  - \$2/kWhr by 2015 (~\$300 for a 5-kg hydrogen system).
  - 3 kWh/kg (9 wt%)
  - 2.7 kWh/L

### **Specific Energy Compared to DOE Goals**



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### **Cost\* Efficiency Compared to DOE Goals**



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\*Based on \$11/Ib Carbon Fiber. Cost Numbers Derived from PNNL Cost Analysis. See Notes.



#### \*Cost Numbers Derived from PNNL Cost Analysis. See Notes.

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- Based on the system cost:
  - Carbon fibers drive 33% of system cost
- Based on tank cost:

**Observations** 

- Raw carbon fibers are 50% of tank costs
- Composite part of tank is 86 to 89% of total cost
- Reduce fiber cost from 11 to \$6/lb would:
  - Reduce tank cost by 24%
  - Reduce system cost by 15%
- Hardware components are expensive: 52% (of system cost)
  - Valves, Regulators, Sensors, End Bosses, Fittings
  - Explore ways of reducing these costs.
- Realistic, aggressive DOE cost goals •
  - Cost Efficiency == \$/KwH ۲
    - \$1700\* for full system
    - Realistic goal = 10 \$/KwH
  - Specific Energy == KwH/kg
    - Driven by fiber performance.
    - Realistic goal == 2 KwH/kg (2010 Goal)
- **Refurbishing tanks** 
  - 30-year life time is longer then vehicle life
  - Use refurbished tanks in lower-cost new vehicles
  - Residual value of tank after 15-years maybe 40% of new
- 1.80 vs. 2.25 Factor of Safety?





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- Reduction in hazardous chemicals and overall waste
- Significant reduction in tooling
- 30-40% reduction in manufacturing assembly time
- Process creates repeatable first-time quality



Much better buy-to-fly ratio

Manufacturing techniques drive environmental performance

# New Head Design

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Ref: Boeing Released, BOE031709-109, by K. M. Nelson on Jan 29, 2010.

- Structural efficiency increases with smaller fiber-placed polar openings
- Heads must be designed to minimize clearance with the boss
- Programming focused on geodesic paths for minimal shear loading of composite







