# Solid Oxide Fuel Cell (SOFC) Technology for Greener Airplanes

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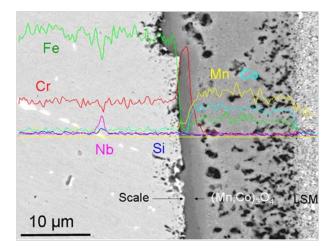
# **SOFC Technology Development at PNNL**

#### PNNL has been active in SOFC development since 1987.



Glycine – nitrate powder synthesis of La<sub>0.7</sub>Ca<sub>0.31</sub>CrO<sub>3</sub>

#### Major participant in SECA Core Technology Program since 2000.



SOFC stack electrochemicalthermal modeling

Store 59 Store 50 Store

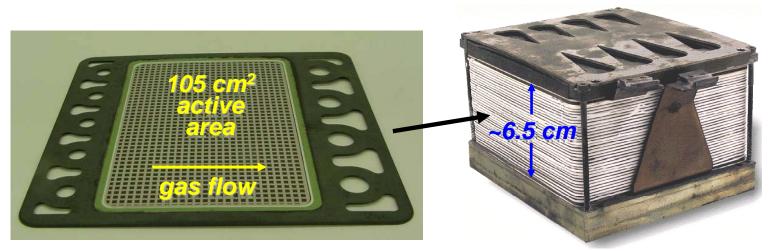
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Development of electrically conductive coatings to prevent Cr volatility from SS

# **PNNL-Delphi SECA Team**

- Transferred SOFC technology to Delphi starting in 1999.
- PNNL now purchases stacks from Delphi for integration into specialty systems.



Single Cell in Co-flow Cassette

30-Cell Stack in Frame 700-2300 watts, 9 kg, 2.5 L



# Delphi is Developing Auxiliary Power Unit (APU) for Long Haul Trucks



Supports "hotel load" so ICE can be shut down at night.

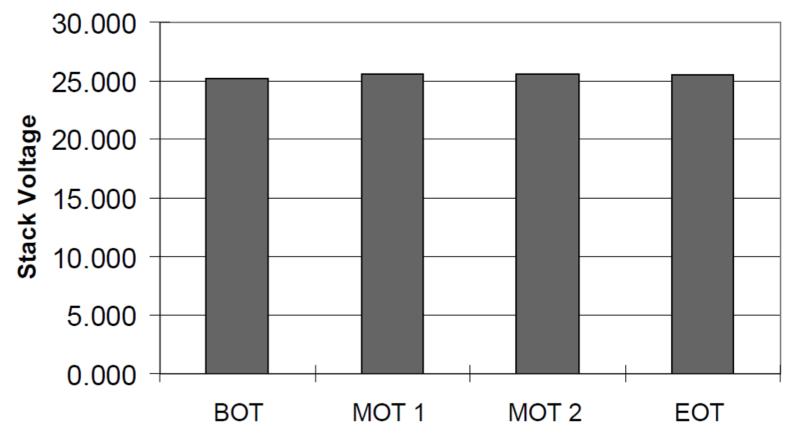
➢Runs on low-sulfur diesel (<15 ppm S)</p>





## Vibration Testing of 30-Cell Stack - Delphi

30-cell stack voltage at 60 Amps (570 mA per cm2)



•Military Standard d 810G Method 514.6 Annex C "Truck and

Transportation Over US Highways"

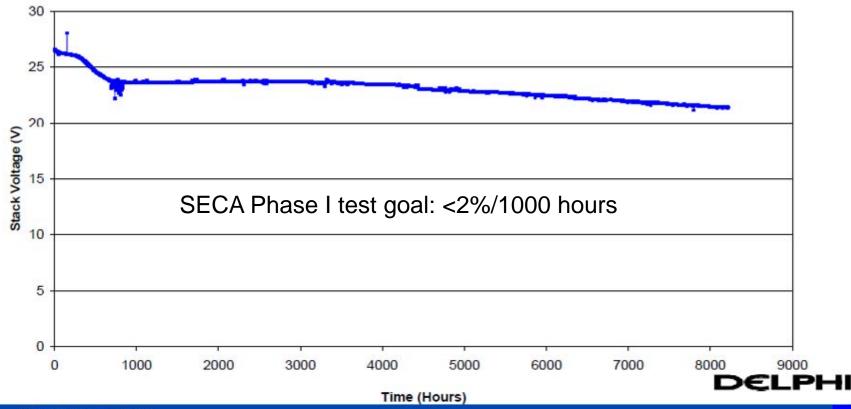
•10-800 Hz, 3 Axes

•Simulated 750,000 miles of driving



#### Gen 3.2 30-Cell Stack Durability

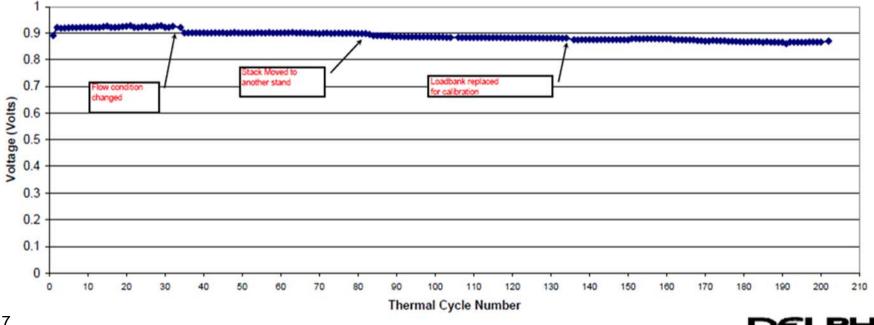
- Fuel = 48.5% H2, 3% H2O, rest N2; current = 333 mA per cm<sup>2</sup>
- > 8200 hours, continuing to run
- Total degradation is 1.20% per 500 hours
- Degradation after initial lowering of power and stabilization is 0.66% per 500 hours
- Implementing solution to mitigate initial lowering of power



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#### Thermal Cycling

- Gen 3.2 30-cell stacks evaluated for thermal cycling with improved seals
- 200 thermal cycles demonstrated with minimal degradation
  - 2 hour heat-up
  - Performance evaluated at each thermal cycle
  - Constant current load of 285 mA per cm<sup>2</sup> at operating temperature
  - Fuel of 48.5% H<sub>2</sub>, 3% H<sub>2</sub>O, rest N<sub>2</sub>





## **Key Advantages of SOFC for Aircraft**



- $H_2$ , CO and CH<sub>4</sub> can be directly supplied to the anode as fuels, so an SOFC system can more easily use liquid hydrocarbons.
- High temperature supports steam reforming, which boosts system efficiency.



### **Efficiency Boost from Steam Reforming**

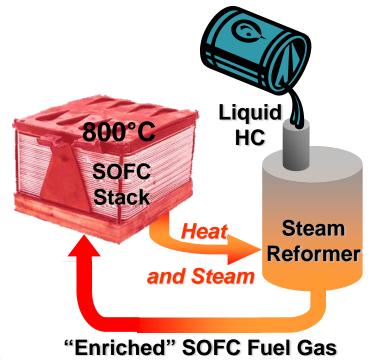
Steam reforming is endothermic

 Heat from SOFC stack is converted into ~25% *increased* chemical energy of reformate:

Steam Reformation of *n*-Dodecane:  $C_{12}H_{26} + 12H_2O + heat \rightarrow 12CO + 25H_2$ 7552 9421 kJ/mole kJ/mole

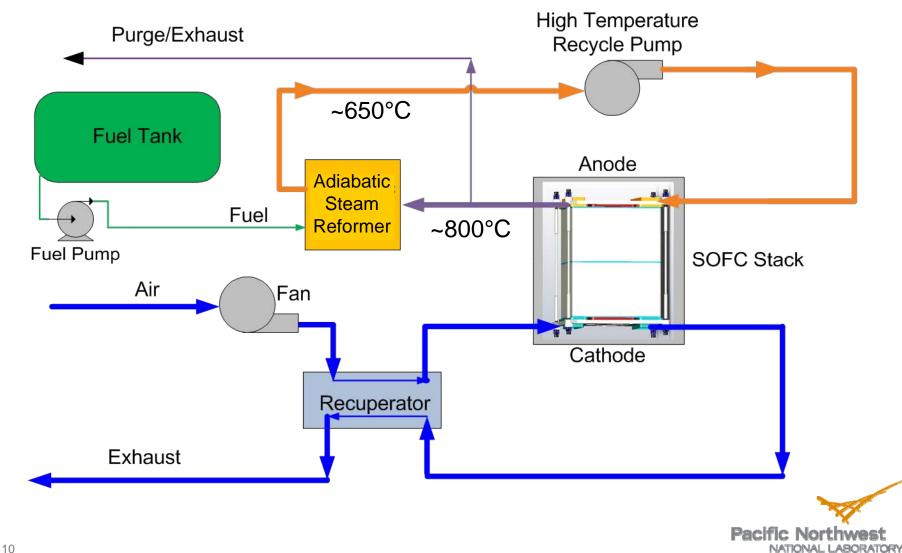
• System can yield 65% net efficiency

 Steam and heat for reforming obtained from SOFC stack exhaust

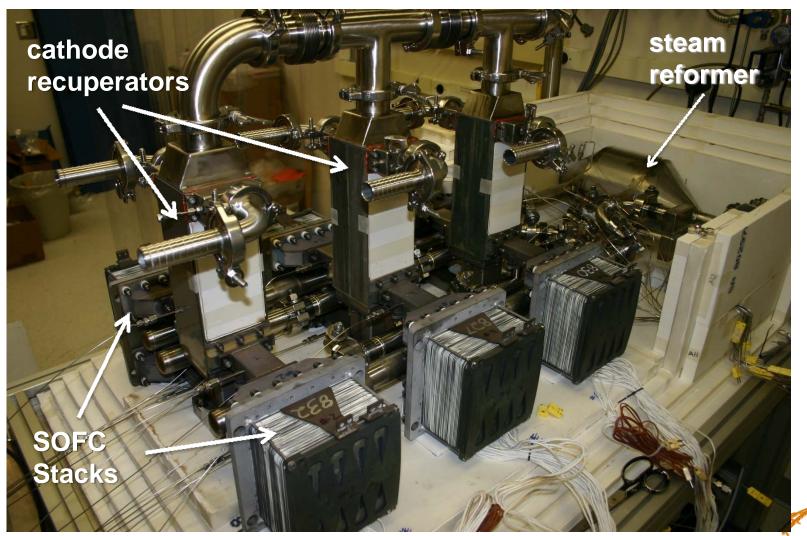




### **SOFC Power System with Steam Reforming and Anode Recycle**



### Demonstration System with Steam Reforming and Anode Recycle



### **Key Challenge for SOFC in Aircraft**





#### Must increase specific power!



## The Numerator: Factors Affecting Power Density

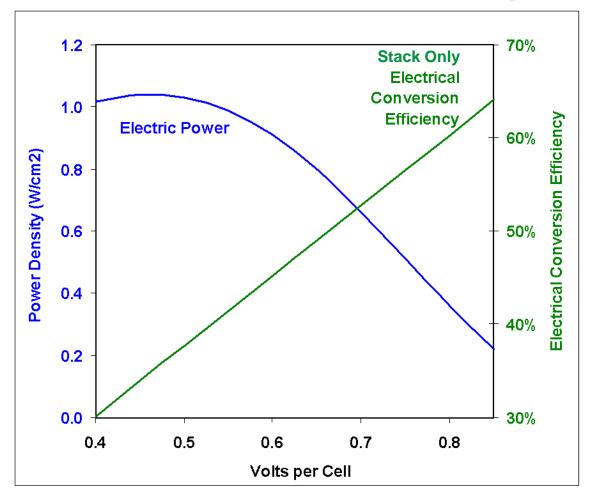
> Voltage

> Temperature

- Fuel Concentration
- Electrochemical Activity



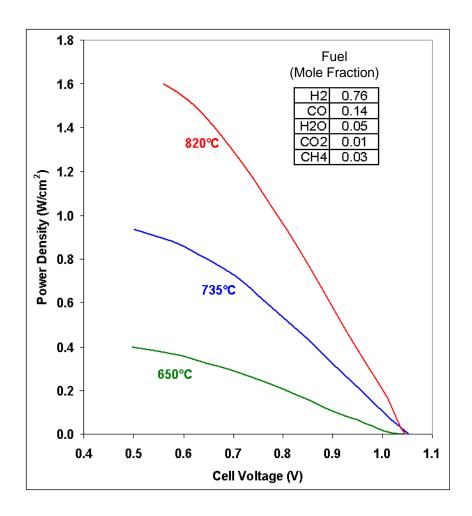
### High Power Density Comes with Lower Efficiency



Operation below ~0.7 volts may accelerate degradation.

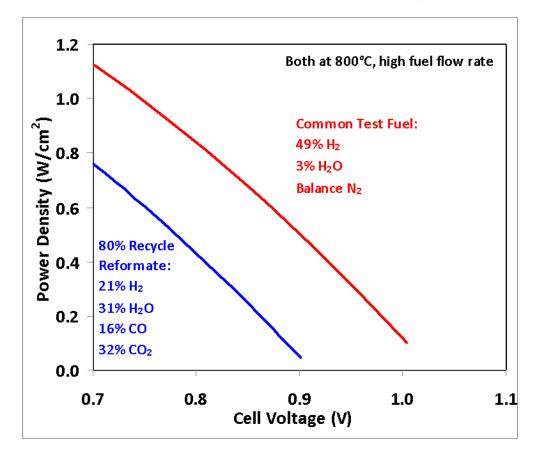


### Upper Limit for Steel Framed Planar SOFC Technology: ~850°C



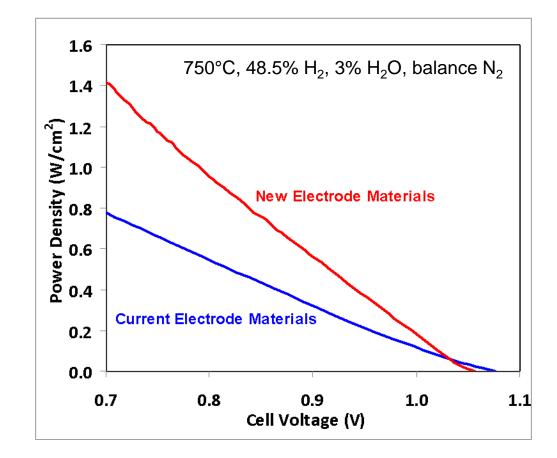


### Reformate Gives Lower Power than Moist Hydrogen





### Better Times are Ahead: Electrochemical Activity is Improving





### **Stack Weight**



Current technology: ~1.5 kW, 9 kg ~ 0.17 kW/kg

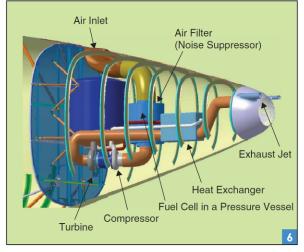
Specific power required for aircraft stack: ~ 1.0 – 1.5 kW/kg



## **System Weight**



Current technology not designed for low mass: 6 kW, ~170 kg ~ 0.035 kW/kg

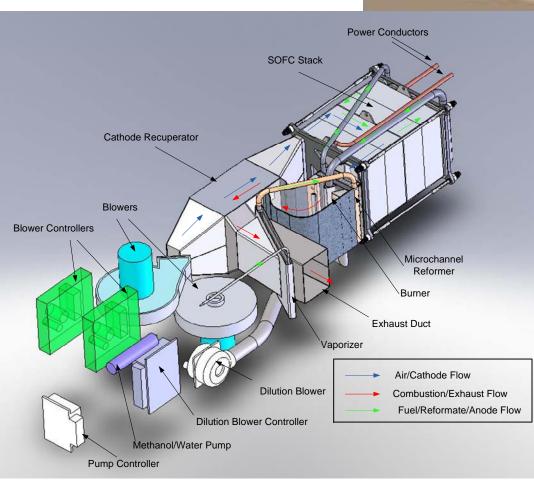


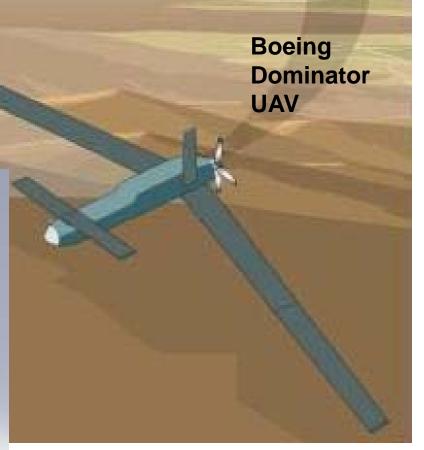
 Specific power required for aircraft system:
~ 0.4 to 0.5 kW/kg



K. Rajashekara, J. Grieve and D. Daggett, IEEE Industry Applications Magazine, July-August, 2008

### Demonstration System for UAV (AFRL)





Short flight, methanol-water fuel

Pacific No

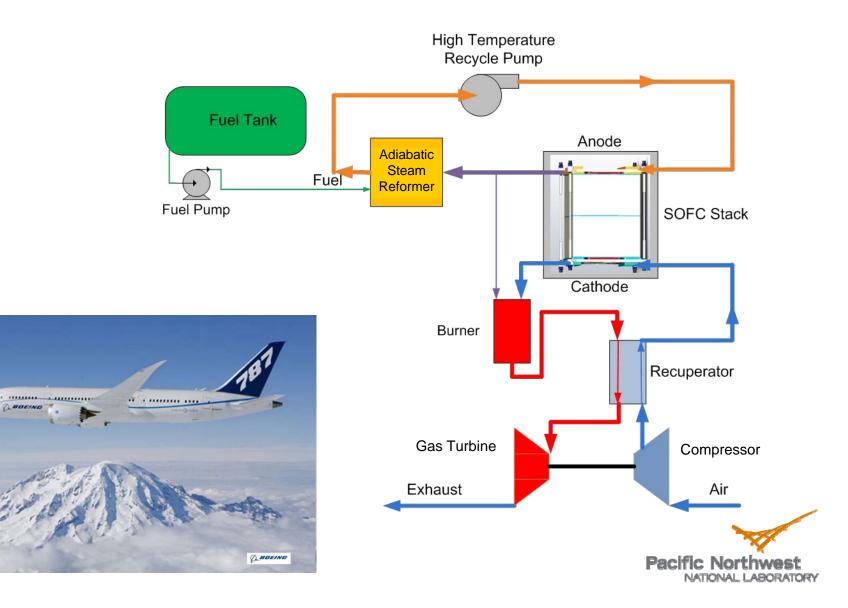
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> 0.55 volts/cell, low efficiency

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> 3 kW, 12.3 kg → 0.24 kW/kg
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## **Possible Configuration for Aircraft**



ABBEING

## **Greener Airplane Study**

Assumptions:

- Reference airplane will be the more electric 787.
- SOFC to supply base load.
- Metrics are overall airplane fuel efficiency and emissions.

#### Questions to be addressed:

Determine load profile for 787 (have made a good start with Joe Briet's help)

- What size and configuration for SOFC base-load system?
  - Combine with turbine-compressor?
  - Run at elevated pressure?
  - How much anode recycle?
  - What voltage (power vs. efficiency)?
  - Two SOFC systems for redundancy?
- What technologies for peak loads?
  - Turn up SOFC
  - PEM, HTPEM
  - Turbine
  - Batteries, ultra-capacitors
  - Combinations



## **Greener Airplane Study, Cont.**

#### Questions, continued:

- De-sulfurization on-board or at airports?
- If PEMFC used for peaking power, how should H2 be supplied?
  - Storage in tanks
  - Metal hydrides
  - Obtain by membrane separation from SOFC system reformate
- Assume water will be condensed and collected. How to use it?
  - Cabin air humidification
  - Lavatories
  - Injection into jet turbines to reduce emissions
- Can waste heat be used for de-icing or...?
- > Should  $CO_2$  be separated for fuel tank inerting?
- Assess benefits of optimum system over existing (787) technology using overall airplane fuel economy and emissions as metrics.
- Assess technology readiness levels and pathway to deployment.



### **Thank you!**



### **Questions?**

